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NATURE OF PROPOSAL:

A. Install ISi Automatic Flight Control System (AFCS) and Attitude Heading Reference System (AHRS) components as listed below:

COMPONENT TITLE	COMPONENT P/N
NOSE SECTION:	
AFCS Computer Trim Coupler Rate Gyro Package (3 Each 2157-30AK-1 or 3 Each 2152-30V-1 Rate Gyro* are included)	11CE019A 1478C 121296-01
Mach Controls Two Gyro Platform AHRS Amplifier & Power Supply AHRS Adapter AHRS Power Converter Control Relays Support Bracketry	1420H-01 2171AB 3311H 6502F 1601F

COCKPIT SEXTION:

AFCS Controller AFCS Gain Pots (3 Each)	11CC108A
AFCS Auto Trim c/o Switch & Light AFCS Auto Trim Test Switch	1
AFCS Pitch Trim Indicator* AFCS Roll Trim Indicator* AHRS Controller	707502-03 Series Qh18-5 707502-01 Series Qh18-4 380hG
Circuit Breakers (2 Each) AHRS Power Control Switch Bearing Distance Heading Indicator	Micro 6ET1-T
Circuit Components Required for new Inverter Control	248745 Model 1163 Type 8L (WSPO ONLY)
Support Bracketry	

*See Footnote on Page 3

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NATURE OF PROPOSAL: (Cont'd)

COMPONENT TITLE COMPONENT P/N

Q-BAY SECTION:

AHRS Rate Switch Gyro* 2159C-01
750 VA Inverter (WSPO ONLY) MGH182-100

Various Circuit Hardware Required for Inverter Circuit (WSPO ONLY)

Support Bracketry

FUSELAGE AFT OF Q-BAY AND OVER FRONT WHEEL WELL

 Pitch Serve
 56004-000

 Yaw Serve
 56004-000

 Pitch Capstan*
 913B-4-01

 Yaw Capstan*
 913B-4-01

FRONT WHEEL WELL:

Roll Servo 56004-000 Roll Capstans 913B-4-01

AFT TAIL SECTION:

Pitch Followup 4000A Yaw Followup 4000A

Support Bracketry & Linkage

RIGHT WING:

Roll Followup 4000A

Support Bracketry & Linkage

*This part will be removed from the aircraft, overhauled and/or modified as required to return the component to a "like new" condition. After overhaul or modification, the part shall have a "-01" affixed to the existing part number. The trim indicators will be identified with a "-1" and "-5" in lies of the "-01" number.

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NATURE OF PROPOSAL: (Cont'd)

LEFT WING

Roll Followup Flux Valves Support Bracketry Followup Linkage

4000A 664543-01 Type C2

VARIOUS CIRCUIT COMPONENTS

Connectors (64)
Relays (9)
Switches (4)
Potentiometers (3)
Circuit Breakers (8)

- B. WIRING TO BE INSTALLED.
 - a. Nose Section complete.
 - b. Cockpit complete (in conjunction with Cockpit Update LAC ECF 164).
 - c. Servo wiring to fuselage over, and in wheel well.
 - d. Followup wiring to each wing and to tail section.
 - e. Flux Valve wiring from Left Wing disconnect to Nose Section.
 - f. Q-Bay wiring for Inverter Installation (WSPO ONLY).

REASON FOR PROPOSAL:

To provide an Automatic Flight Control System and Attitude Heading Reference System that:

- a. is designed and manufactured in accordance with current state of the art practice.
- b. has performance characteristics compatible with the required aircraft mission profile.
- c. provides increased system and component reliability.

*See Footnote on Page 3

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DISPOSITION OF SPARES AFFECTED:

After all Articles have been modified the following Components will have no further requirement on this aircraft. A suitable quantity of Spare Components should be maintained until all aircraft have been turned around.

COMPONENT TITLE		PART NUMBER	
Controller Servo (3 Each) Followup Followup (3 Each) Mach Amplifier Trim Control Elapsed Time Indicator A/P Control Gyro 500 VA Inverter (WSPO ONLY) Compass Gyro & Amplifier Control Panel Transformer Indicator	(PROJECT) (WSPO)	996H 118BFX or 118BF 911A-1 1450D-1 119611-01-01 113459-02 Type MH-1 114705-02R345-2 114705-02R338-2 R419-2 2156A & 3303A 3801A 1611A 1D250A/ARN	or

VARIOUS CERCUIT COMPONENTS

Resistors (6)
Capscitors (5)
Filters (3)
Connectors (64)
Relays (12)
Switches (7)
Potentiometers (8)
Circuit Breakers (7)

WEIGHT & EALANCE:

There is no increase in weight. However, the equipment is now installed in the nose of the aircraft and moves the C.G. FORWARD. To maintain a zero C.G. shift (ten) 10 additional pounds of ballast must be installed at F. S. 673.